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Authors: Sofyan Djamil, NPG Suardana, Agustinus Purna Irawan and IKG Sugita

Corresponding Author: agustinus@untar.ac.id

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Prof. Dr. Ir. Agustinus Purna Irawan

**Professor of Mechanical Engineering** 

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# International Journal of Engineering Research and Technology (IJERT)

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#### **Contents**

Characterization of Calcia Stabilized Zirconia Synthesis for Solid Oxide Fuel Cell Electrolytes through Precipitation Method

pp. 4011-4015

M. Nurbanasari, D.G. Syarif, M.J. Fahmy, Y. Irwan and A.P. Siswanto

Tower Wake Distortion Effect: A Comprehensive Review of Methods and Applications

pp. 4016-4032

Maduako E. Okorie and Freddie Inambao

Energy Audit On Primary Municipal Facilities: Reflection of Municipality's Energy Consumption as a Direct Consumer of the Energy Utility (Eskom)

pp. 4033-4047

Melusi Nhleko and Professor Freddie L. Inambao

Role of Sense of Place in the Use of Communal Spaces as Places for Social Interaction at an Owned Low-Cost Flats Bendungan Hilir II

pp. 4048-4064

Joni Hardi, Prof. Liliany Sigit Arifin and Yohanes Basuki Dwisusanto

Membrane Modeling and Simulation for a Small Scale Reverse Osmosis Desalination Plant

pp. 4065-4083

Randy Ncube and Professor Freddie L. Inambao

**Trends: Energy Efficiency and Energy Security** 

pp. 4084-4117

Jerusha Joseph and Freddie L. Inambao

#### **Application of Resampling Techniques in Orthogonal Regression**

pp. 4118-4124

Anwar Fitrianto, Tan Sin Yun and Wan Zuki Azman Wan Ahmad

# Experimental Analysis of a Thermoelectric-(Vapor Compression) Hybrid Domestic Refrigerator

pp. 4125-4133

Yasser Abdulrazak Alghanima, Osama Mesalhy and Ahmed Farouk Abdel Gawad

A New Passive-Active Method of Protection from Dynamic Vortex Atmospheric Structures: Physical Foundations, Technical and Economic Advantages

pp. 4134-4138

M.E. Romash, A.Y. Varaksin and M.V. Protasov

Transformation Space Due To Conflict (A Case Study of Kuta, Bali, Indonesia)

pp. 4139-4145

Agung Wahyudi, Imam Buchori, Joesron Ali Syahbana and C. Widi Pratiwi

#### **Dual Scalar Aharonov-Bohm Effect and the Photon Mass**

pp. 4146-4150

Maribel C. P'erez Pirela and Miguel E. Rodriguez R. y

#### Accuracy Improvement Technique of Big Data based LBS System

pp. 4151-4155

Changbae Mun

Dynamic Shear Rheometer to Measure the Improvement of Asphalt Properties with the Addition of Buton Natural Asphalt-Rubber (BNA-R)

pp. 4156-4162

Sigit Pranowo Hadiwardoyo, R. Jachrizal Sumabrata and Nurul Wahjuningsih

Modeling the Impacts of Liquid Entry Pressure on Membrane Performance during Oil-Water Separation

pp. 4163-4170

P. B. Sob

The Effect of Spelling with User's Mother Tongue on P300 Speller Performance

#### **Evaluation of Using Standard Mobile OFDM Signals for Short-Range Radar Sensors**

pp. 5000-5004

Elena Omelyanchuk, Andrey Tikhomirov, Ilya Muraviev, Niek Molenkamp and Olga Simonova

#### Development of a Monitoring System Automatic Power Meter at the "Gunung Salak" Geothermal Power Plant, Sukabumi, Indonesia

Muhamad Muslih. Nunik Destria Arianti and Somantri

#### A DEVELOPMENT OF A SYSTEMATIC IN-SILICO ANALYSIS FOR RNA-Seq ANOPHELES GAMBIAE DATA

pp. 5012-5027

Marion Adebiyi, Samuel Oladayo Olawepo and Micheal Olaolu Arowolo

#### **Energy and Data Communication Delay Aware Routing in WSN**

pp. 5028-5032

G. Vishnupriya and R. Ramachandran

#### Evaluation on the Effectiveness of Priority Seats for the Pregnant Women on Railroad Vehicles in South Korea

pp. 5033-5046

Mizuno Tomomi and Tokuda Katsumi

#### Ballistic Failure Mode of Apus Bamboo Strips Reinforced Epoxy Composite Materials

pp. 5047-5050

Sofyan Djamil, NPG Suardana, Agustinus Purna Irawan and IKG Sugita

#### MAC-Based Physiological signal extraction approach for WBAN on Body Postures

pp. 5051-5054

A.Roshin and K.V.D.Kiran

#### Integrating Dehydration and Natural Gas Liquids Processes for Maximization of Natural Gas Liquids Production

pp. 5055-5065

Ahmed Ali Masnsour, Walaa Mahmoud Shehata and Fatma Khalifa Gas

#### Performance of a Split Bregman Method for a TVL1-type of Image Restoration Model

pp. 5066-5077

Hyo Jin Lim and Jae Heon Yun

#### **Combined Approach for Treating Stochastic Vector Optimization Problem**

pp. 5078-5082

Ahmed A. Elsawy, Adel M. Widyan and Atheer S. Alqudhaibi

#### Effect of Piezoelectric Thickness Ratio on the Deflection of Laminated Hybrid Composite Plates

pp. 5083-5094

D. Dhanunjaya Raju and V. V. Subba Rao

### Quality Management Systems (QMS) and Organizational Performance

pp. 5095-5104

Juan Manuel Andrade, Gerardo Duque Gutierrez and Fernando Fierro Celis

#### Investigating Factors that Affect Purchase Intention of Visitors of E-commerce Websites Using a High Scoring Random Forest Algorithm

pp. 5105-5112

Martha Teiko Teye and Yaw Marfo Missah

#### Hole Filling And Image Fusion Approach For RGBD Database

pp. 5113-5122

Aniketh A. Gaonkar, Narayan T. Vetrekar and Rajendra S. Gad

#### **Enabling Financial Inclusion By Technology Led Last-Mile Delivery Of Banking Services**

pp. 5123-5128

Govind Korekar and Dr. Anuja Agarwal

#### Finding Accuracy of Utterance of Language Model

pp. 5129-5134

Nadeem Ahmed. Kanasro, Najma Imtiaz Ali, Ghulam Muhammad, Mujeeb U Rehman Maree and A.G Memon

#### Spread of Avian Influenza- A Mathematical Approach

pp. 5135-5141

Bhanu Sharma, Pooja Khurana and Deepak Kumar

#### A Content-Based Retrieval Model with Combinational Features and Indexing for Distributed Video Objects

Nithya Kaliaperumal, Akansha Das and Vijayakumar Balakrishnan

#### **Design Challenges of Securing IoT Devices: A survey**

pp. 5149-5165

Matasem Saleh, NZ Jhanjhi, Azween Abdullah and Raazia Saher

#### **Evolution of Smart Grid Assessment Methods: Science Mapping and Performance Analysis**

pp. 5166-5175

Eliseo J. Zarate P., Ana. L. T. S. Da Motta and Juan H. Grados G

# Ballistic Failure Mode of Apus Bamboo Strips Reinforced Epoxy Composite Materials

Sofyan Djamil<sup>1</sup>, NPG Suardana<sup>2</sup>, Agustinus Purna Irawan<sup>3</sup> and IKG Sugita<sup>4</sup>

<sup>1,3</sup>Mechanical Engineering Department, Universitas Tarumanagara, Indonesia <sup>1,2,4</sup>Mechanical Engineering Department, Universitas Udayana, Indonesia

#### **Abstract**

Indonesia has abundant natural fibre potential to be developed into engineering materials with good strength. One of the potential natural strips in Indonesia is bamboo fibre. This study aims to determine the tensile strength, flexural and ballistic failure mode of Apus Bamboo Strips Reinforced Epoxy Composites Materials (BSRECM) as an alternative material for making bulletproof products. The research was carried out by making epoxy bamboo fibre composites with laminate configurations of  $0^0/0^0/0^0$  and  $0^0/90^0/0^0$ , thickness of 4.5 mm, 3 layers, and volume fraction of 30%. Bamboo strips were soaked in 5% NaOH, for 72 hours, and dried for 48 hours at room temperature. Following that process, bamboo strips are heated in an oven at 40°C for 2 hours, then woven and laminated using epoxy resin. The tests carried out were tensile, flexural and ballistic tests. Ballistic testing was carried out with a firing range of 25 m and bullet speed of 900  $\pm$  30 m/s with an Indonesian SS1 rifle. Based on the tests results, it was obtained that the average tensile strength was  $50.49 \pm 3.77$  MPa, the average strain was  $5.5 \pm 0.8\%$ , the average flexural strength was  $54.72 \pm 6.85$  MPa. The ballistic failure mode of epoxy bamboo fibre composites with laminate of  $0^0/0^0/0^0$ configuration in the form of partial penetration failure mode and has a complete penetration mode for specimens with 0<sup>0</sup>/90<sup>0</sup>/0<sup>0</sup> laminate configuration. SEM observations with  $0^{0}/0^{0}/0^{0}$  laminate configuration shows that the bamboo strips are not completely but only locally damaged, the laminate configuration  $0^0/90^0/0^0$  damage that occurs on the front surface, the level of damage tends to widen due to the high impact load of the bullet projectiles. The results of this study will be able to be used as a reference for further.

**Keywords:** Bamboo Strips, Tensile Strength, Flexural Strength, Ballistic Test, Failure Mode

#### I. INTRODUCTION

The development of natural fibre composites continues to increase in proportion with the need for products that are environmentally friendly, recyclable and abundant. One of the natural fibres that is potential to be developed as a technical material with various uses is the Apus bamboo fibre. Apus bamboo fibre is feasible to be developed because it has several advantages such as environmentally friendly, cheaper manufacturing costs and materials, abundant, and can be replanted for sustainability [1], [2], [3], [4], [5].

This study aims to determine the ability of bamboo strips composite with an epoxy matrix as an alternative material for making bullet-proof products. This is in proportion with the increasing need for using bullet-proof protective materials. In making bulletproof materials, it is necessary to design a material that is lightweight and has a high level of protection [6], [7]. Bamboo strips composite is one alternative for making body armour from sharp weapons and bullet shots. Body armours can prevent penetration and the occurrence of defects or damage to the human body [6], [8], [9]. The characteristics of a human body armour can be measured in terms of the interaction between the armour and the human body. Effectively as a protector of the human body, there must be no penetration inside of the human body when receiving external loads such as bullets and sharp weapons.

#### II. MATERIAL AND METHOD

#### **II.I Bamboo Strips Preparation**

The bamboo strips used in this study were Apus Bamboo (Giganchtoloa Apus), taken from Pebasiran village, Parung Panjang, West Java, Indonesia. The bamboo is cut manually using a strip knife with a thickness of 0.5 - 0.7 mm and a width of 5 - 7 mm. The bamboo strips were soaked in 5% NaOH for 72 hours and dried for 48 hours at room temperature (Fig. 1). The bamboo fibres are then heated in an oven at a temperature of  $40^{\circ}$  C for 2 hours to remove the moisture content in the fibre and are made woven (Fig. 2)



Fig. 1. Bamboo Strips Preparation



**Fig. 2.** Bamboo Strips Drying Process at a Temperature of  $40^{\circ}$ C

#### **II.II Fabrication of Laminate Composites**

The manufacturing process of bamboo strips composite is carried out by hand layup method using a mould of  $200 \times 200$  mm steel plate made in 3 layers and 4.5 mm thickness. The matrix used is epoxy resin of diglyceryl ether of bisphenol type A with a density of  $1.15 \times 103 \text{ kg/m}^3$  and working temperature of  $120^{\circ}$ C. The epoxy has the characteristics of having high strength, good dimensional stability, low shrinkage, and a high level of adhesive. Two types of the test samples were made with the orientations of  $0^{\circ}/0^{\circ}/0^{\circ}$  and  $0^{\circ}/90^{\circ}/0^{\circ}$  (Fig. 3)





Fig. 3. Fabrication of the BSRECM

#### **II.III Tensile Strength Test**

In reference to the ASTM D 3039 standard, tensile testing was conducted by using the Universal Testing Machine in a room temperature of 22°C and relative humidity of 63%. The crosshead speed for tensile testing was 2 mm/min. The dimensions of the specimens for tensile testing were 250 x 25 x 4 mm (length x width x thickness). Six samples were tested for tensile testing. The average values were reported including the standard deviations. The Universal Testing Machine and the process of tensile testing are shown in Fig. 4. Each sample was loaded to failure. Based on tensile testing will be obtained for tensile strength and Young's Modulus of the BSRECM [10].



Fig. 4. Tensile Specimen Test of the BSRECM

#### **II.IV Flexural Strength Test**

In reference to the ASTM D 790-03 standard, flexural testing was conducted by using the Universal Testing Machine with a testing room temperature of 22.5°C and relative humidity of 63.5%. The load was applied midway between the supports with a crosshead speed of 1.7 mm/min (Fig. 5). Each sample was loaded to failure. The dimensions of the specimens for flexural testing were 120 x 10 x 4 mm (length x width x

thickness). Six samples were tested for flexural testing. The average values were reported including the standard deviations [10].



Fig. 5. Flexural Specimen Test of the BSRECM

#### **II.V Ballistic Test**

Ballistic testing was carried out at the ballistic testing laboratory of the Indonesian Army Research and Development Agency (Dislitbangad), Bandung, Indonesia, as in Figure 6. This test was conducted by the use of 5.56 mm bullets (MU5-TJ) with an Indonesian-made SS1 gun by the serial number of JaT 97045854, bullet speed  $900 \pm 30$  m/s, firing range of 25 m. Observations on ballistic test results focused on the failure mode that occurred in the BSRECM.





Fig. 6. Ballistic Testing Process of the BSRECM

#### III. RESULT AND DISCUSSION

# **III.I Tensile Test Results**

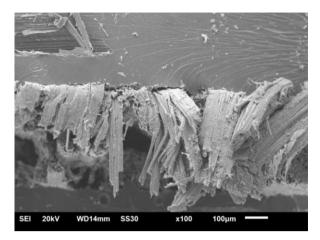
Based on the results of tensile test on the test sample of the BSRECM, the average tensile strength is obtained as follows (Table 1).

Table 1. Tensile Test Results of the BSRECM

Load	Area	Strain	Ultimate tensile strength (MPa)
(kgf)	(mm²)	(%)	
$265 \pm 24.4$	$53.8 \pm 3.7$	$5.5 \pm 0.8$	$50.49 \pm 3.77$

The average tensile strength of the BSRECM resulting from this test is relatively decent. Based on the obtained tensile strength, the BSRECM was assessed further in ballistic testing to obtain alternative materials for making bullet-proof components. The results of the tensile strength are also supported by the results of observations using SEM on the tensile test sample. Based on the SEM test, the interface

between the strips and the matrix is satisfactory, no voids and delamination are seen (Fig. 7).



**Fig 7.** SEM micrographs of the BSRECM laminate composite (100x)

#### **III.II Flexural Test Results**

The average flexural strength of the BSRECM is relatively decent ( $54.72 \pm 6.85$  MPa) as shown in Table 2.

**Table 2.** Flexural Test Results of the BSRECM

b	h	L	Load	Flexural
(mm)	(mm)	(mm)	(kgf)	strength (MPa)
11.65 ±	4.27 ±	63.15 ±	12.95 ±	$54.72 \pm 6.85$
0.54	0.09	0.06	1.51	

Flexural strength is required to withstand the movement of the bullet when fired in ballistic tests. Based on the flexural strength obtained, it is highly possible for the BSRECM to be tested further as an alternative material for making bulletproof components.

#### **III. III Ballistic Test Results**

Ballistic testing aims to observe the response of the BSRECM in receiving a bullet load at a certain speed. The response that occurs can be observed from the form of failure mode due to the load from the bullet fired at a certain distance and speed. Ballistic testing was conducted using SS1 rifles with 5.56 mm calibre bullets (MU5-TJ), a firing range of 25 m, and a BA 04 S type bullet velocity gauge. The test sample was a  $0^{0}/0^{0}/0^{0}$  laminate configuration, 3 layers and volume fraction of 30 %. Ballistic test results are shown in Fig. 8.

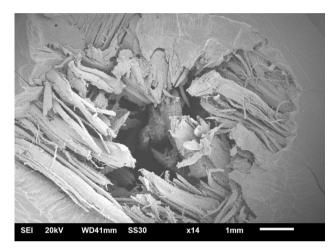






a. Bullets b. Front View c. Rear View **Fig. 8.** Ballistic Test Results of the BSRECM  $0^0/0^0/0^0$  Laminate

Fig. 8 shows the ballistic test results of the BSRECM  $0^0/0^0/0^0$  laminate configuration and bullet velocity of 894.2 m/s. The front and back views of the specimens show the partial penetration failure mode in the form of small deformations around the area of damage. In Fig. 9, the SEM observations on the front surface of the ballistic test specimens with the BSRECM  $0^0/0^0/0^0$  laminate configuration show the fracture area of the front surface as a result of the initial impact load, which occurs in the front surface of the BSRECM. Most of the load strikes the fibres on the bamboo strips, causing damage which causes failure. However, due to the layer configuration, which allows low damage rates, the bamboo strips are not completely but only locally damaged.



**Fig. 9.** SEM Result of  $0^{0}/0^{0}/0^{0}$  BSRECM Laminate after Ballistic Test

The results of ballistic testing conducted with the use of SS1 rifles with 5.56 mm calibre bullets (MU5-TJ), a firing range of 25 m, and a type BA 04 S bullet velocity gauge. The test sample is the BSRECM 0<sup>0</sup>/90<sup>0</sup>/0<sup>0</sup> laminate configuration, 3 layers and volume fraction of 30%. Ballistic test results are shown in Fig. 10.





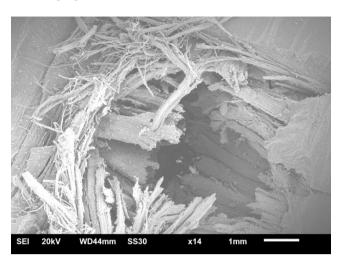
a. Front view b. rear view

Fig. 10. Ballistic Test Results of the BSRECM 0<sup>0</sup>/90<sup>0</sup>/0<sup>0</sup>

Laminate

The greatest degree of deformation at the front surface occurs in the BSRECM with the laminate configuration of  $0^0/90^0/0^0$ . The failure mode describes the occurrence of surface deformation in the front and rear-view specimens caused by the projectile energy. This indicates a complex penetration failure mode. A bullet that is fired into the target area of the specimen results in penetration of the deformed front surface. The type of laminate configuration and the velocity of each projectile can affect the damage and performance of the test specimen [6], [8], [9]. This has been demonstrated by previous researchers who

found that the area of damage increases when the impact load increases [11].



**Fig. 11.** SEM Result of the BSRECM 0<sup>0</sup>/90<sup>0</sup>/0<sup>0</sup> Laminate after Ballistic Test

Fig. 11 shows SEM observations with a laminate configuration of  $0^0/90^0/0^0$  and the damage to the front view surface, due to impact loads. Part of the load strikes the bamboo strips, causing surface damage. The damage occurs on the front surface with the level of damage tends to widen. This is due to the factor of velocity and laminate configuration. The bamboo strips were broken randomly and spread outward from the area of damage. From the condition of the impact load, it is clearly visible on the front surface that the fibre is damaged on the bamboo strip due to the high impact load of the bullet projectiles. The different layers show that the fibre breaks randomly along at its weakest point [12]. The damage that occurs is usually confined to the zone near the impact and is limited by the velocity of the bullet and the impact load. The magnitude of the impact load is one of the factors that contribute to the fibre breakdown [10], [12], [13]. The fibres on the surface which are subjected to impact loads, have been observed to appear with a smooth surface between the laminate boundaries [12], [13]. The results of this study indicate that the BSRECM is exceptionally potential to be further developed as an alternative component for bullet-proof materials. More testing is needed with various angles of fire, ballistic speeds and ranges of fire. These data are very useful for the further design of epoxy bamboo fibre composites as a bullet-proof material.

#### IV. CONCLUSION

Research and ballistic testing have been carried out on epoxy bamboo fibre composites. The observation focus is the failure mode of the composite material against the impact load of a bullet fired at a 25 m radius. Based on this research, it can be concluded that specimens with a  $0^0/0^0/0^0$  laminate configuration have a partial penetration failure mode and the specimen with a  $0^0/90^0/0^0$  laminate configuration has a complex penetration mode of failure. SEM observations with  $0^0/0^0/0^0$  laminate configurations showed that the bamboo strips were not completely but only locally damaged. Furthermore, the results of the  $0^0/90^0/0^0$  laminate configuration indicate damages to the front surface, with the degree of damage tending to

expand. From the perspective of the impact load, the bamboo strips were damaged on the front surface due to the high impact loads of the bullet projectiles. The results of this study will be able to be used as a reference for further development.

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